

# Wireless Attacks on Aircraft Instrument Landing Systems

Harshad Sathaye, Domien Schepers, Aanjhan  
Ranganathan, and Guevara Noubir

Presentation by Colton Asnes



# Table of Contents

## **Background**

How ILS systems work

## **Attacks**

Attack vulnerabilities; Experiment set up and results

## **Review**

Acceptance/Rejection of paper;  
Discussion

# Background

Everyday approximately 5,000 flights are in the air at any given time

- Atlanta's Airport handles around 2500 takeoffs and landings each day

Air traffic controllers rely on wireless technologies

- Communication with pilots via VHF
- Automatic Dependent Surveillance Broadcast (ADS-B)
- Traffic Alert and Collision Avoidance System (TCAS)
- Aircraft Communications Addressing and Reporting System (ACARS)
- Navigation aids such as GPS, Non-directional radio beacons (NDB), Distance Measuring Equipment (DME), and Instrument Landing System (ILS)



# Current Documented Vulnerabilities

01

## **ADS-B Spoofing**

Injecting non-existing aircraft into the sky

02

## **ADS-B Jamming**

Can modify the route of victim aircrafts

03

## **Unsecure ACARS**

Leaked personal passenger info, including medical/financial

04

## **TCAS Spoof**

False resolution advisories that force the pilot to conduct avoidance maneuvers

# Goal of the paper

To investigate the  
vulnerability of aircraft  
instrument landing systems  
(ILS) to wireless attacks

# Instrument Landing System (ILS)

59% of fatal accidents in flight occur during descent, approach and landing (Boeing)

- First deployed in 1932 in Berlin
- ILS provides horizontal and vertical guidance
- FAA typically does not allow landing using visual estimates

Nearly 95% of landings at Boston Logan - essential to evaluate its vulnerability to cyber-physical attacks

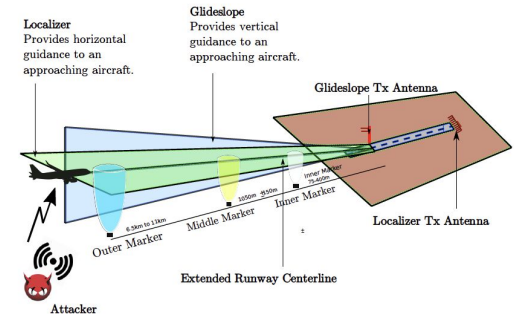


Figure 1: Overview of ILS sub-systems. The ILS consists of three subsystems: i) Localizer, ii) glideslope, and (iii) marker beacons.

# Instrument Landing System (ILS) continued

Three independent subsystems:

1. Localizer
2. Glideslope
3. Marker beacons

In total, provides aircraft with horizontal and vertical guidance to indicate the distance to the desired point of landing

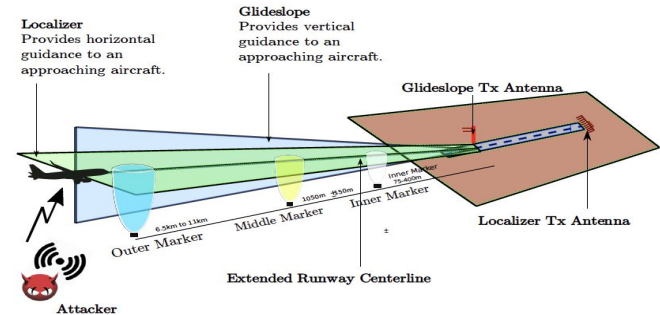
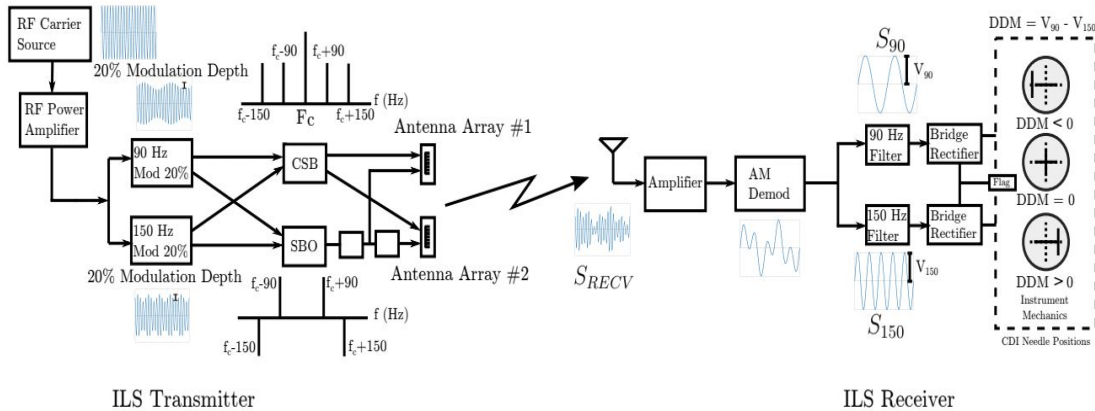


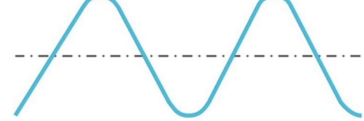
Figure 1: Overview of ILS sub-systems. The ILS consists of three subsystems: i) Localizer, ii) glideslope, and (iii) marker beacons.

# ILS Transmitter and Receiver

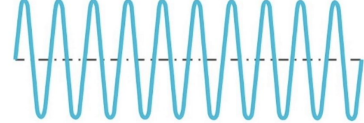


## Amplitude Modulation (AM)

Input (Modulating Wave)



Carrier



Modulated Result





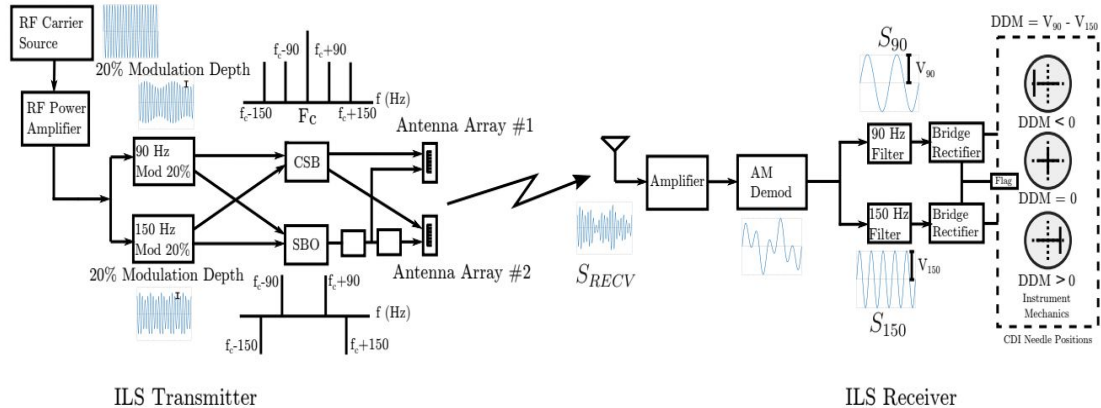
# Instrument Landing System (ILS) continued

ILS Receiver:

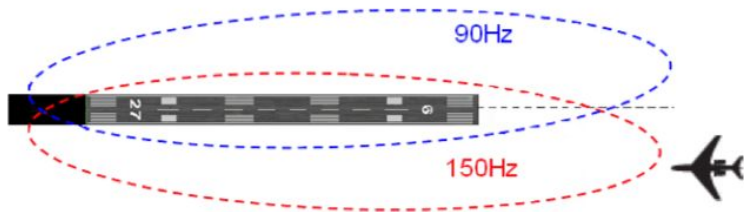
Amplified, demodulated,  
converted to recover DC voltage  
levels

Directly proportional to depth of  
modulation

Full scale deflection:  $DDM > 0.155$  or  
 $DDM < -0.155$  for localizer and if  $DDM$   
 $> 0.175$  or  $DDM < -0.175$  for  
glideslope



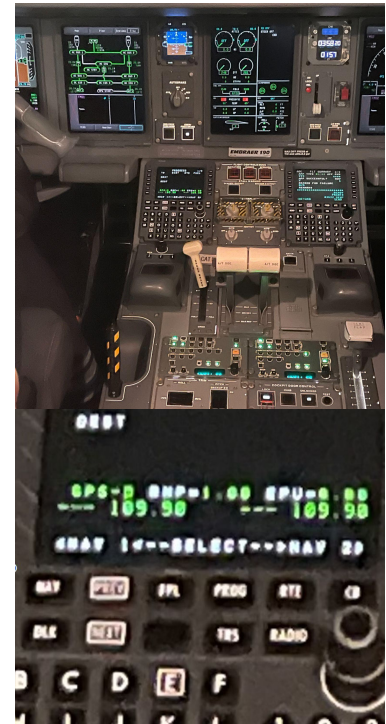
# Approach scenario



Pilot enters localizer frequency into autopilot.



ILS receiver intercepts localizer



Pilots make adjustments

# Wireless Attacks on ILS



## **Overshadow Attack**

Specially crafted ILS signals



## **Single-Tone Attack**

Transmission of one sideband tone

# Overshadow Attack

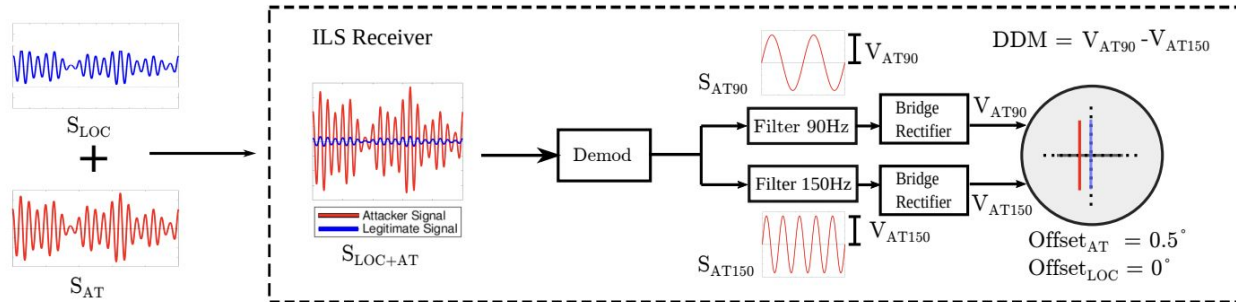


Figure 3: Schematic of the overshadow attack. The attacker's signal has a preset DDM corresponding to  $0.5^\circ$  to the right of the runway. Attacker's signal overshadows the legitimate signal. The blue line represents the needle position without attack.

# Overshadow Attack continued

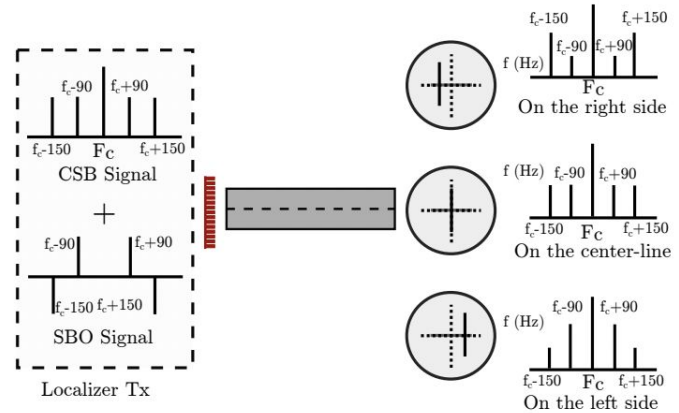
Amplitude is dependent on the position of the aircraft relative to the runway and its approach path angle

$$\text{DDM} = \text{VLOC90} - \text{VLOC150}.$$

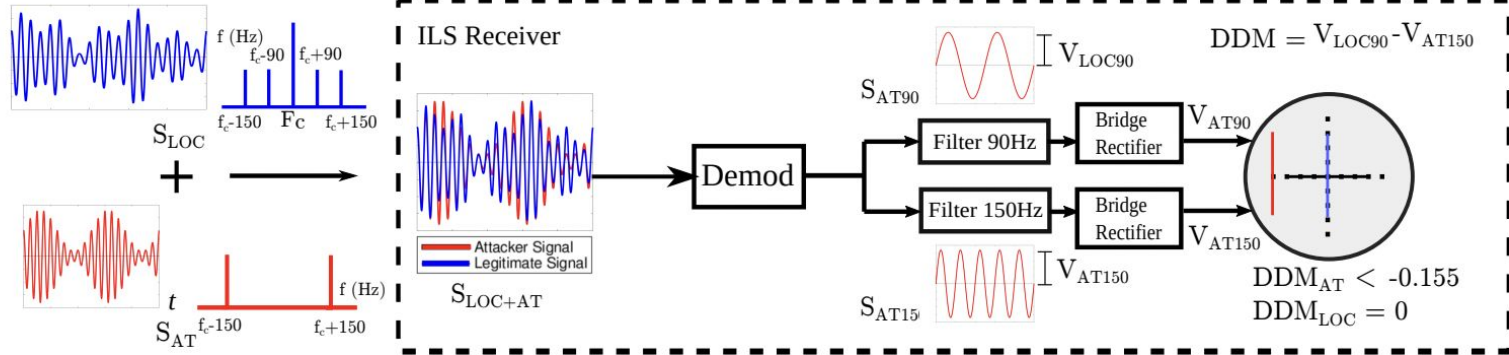
With the attacker spoof, the new depth is:  
 $[\text{VLOC90} + \text{VAT90}] - [\text{VLOC150} + \text{VAT150}]$ .

Since  $\text{VAT90} \gg \text{VLOC90}$  and  $\text{VAT150} \gg \text{VLOC150}$

the resulting  $\text{DDM} = \text{VAT90} - \text{VAT150}$



# Single-Tone Attack



# Experimental Setup

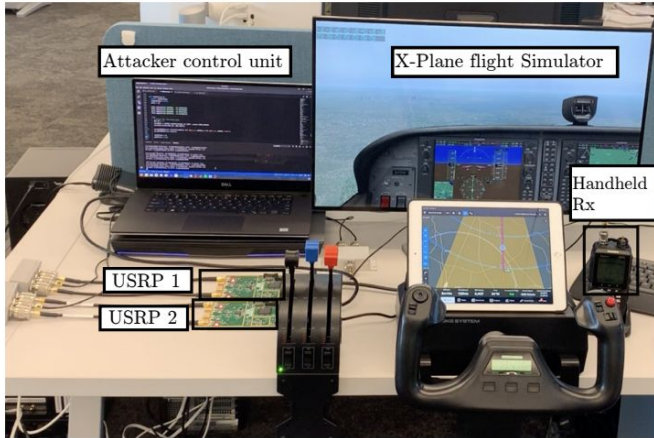
**X-Plane 11 flight simulator**

**Attacker control unit**

**USRP B210s**

**Commercial aviation grade handheld navigation receivers: Yaesu FTA-750L and Sporty's SP-400 Handheld NAV/COM Aviation**

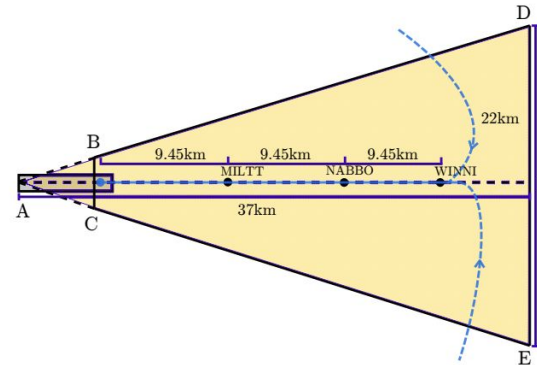
# Attacker control unit



## Spoofer Zone Detection

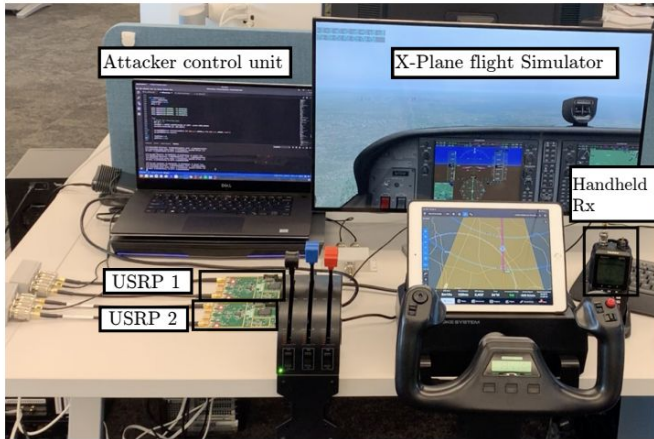
Automated and timely triggering of the spoofing signal

Even-Odd algorithm in scalable vector graphics (SVG)





# Attacker control unit



# Offset Correction Algorithm

Current offset =  $\theta = \tan^{-1}[(mCA - mBA) / (1 + mBA * mCA)]$ , where  $m$  is the slope

DDM =  $(DDM_{fullscale} * \Delta) / 2.5$  (2.5 = angle for full-scale deviation)

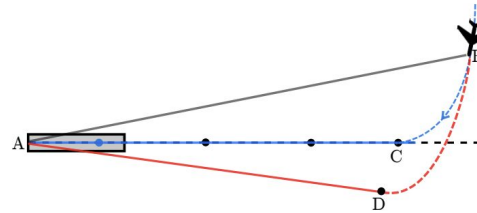
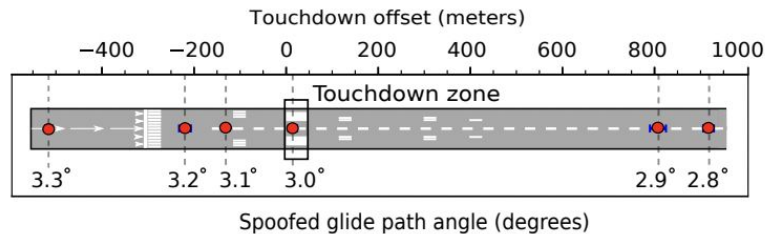
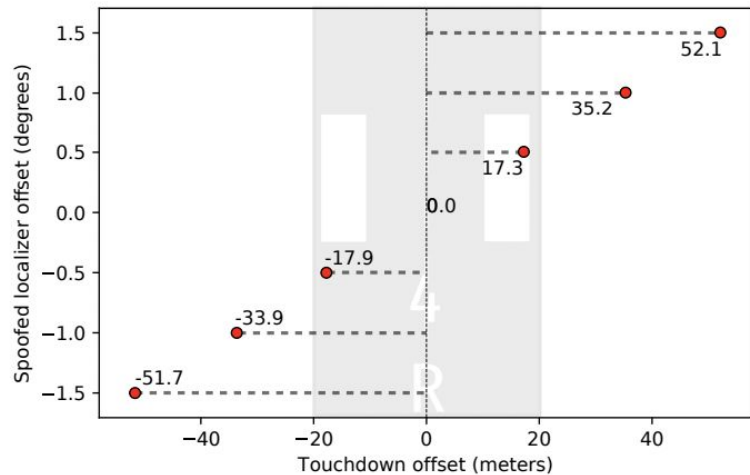


Figure 11: Offset correction algorithm takes into account aircraft's current position to calculate the difference in the spoofed offset and the current offset.

## Algorithm 1 Offset correction algorithm.

- 1: **procedure** GETANGLEDIFFERENCE
- 2:    $\angle DAC \leftarrow TargetedLocalizerOffset$
- 3:    $\angle BAC \leftarrow GetAngle(location)$
- 4:    $difference \leftarrow \angle DAC - \angle BAC$
- 5:   **return**  $difference$
- 6: **procedure** CALCULATEDDM
- 7:    $difference \leftarrow GetAngleDifference$
- 8:    $ddm \leftarrow (0.155 * difference) / 2.5$
- 9:    $AT90 \leftarrow 0.2 + (ddm) / 2$
- 10:    $AT150 \leftarrow 0.2 - (ddm) / 2$
- 11:   **ChangeAmplitude**(AT90, AT150)

# Evaluation of Overshadow Attacks



# Evaluation of Single-Tone Attacks

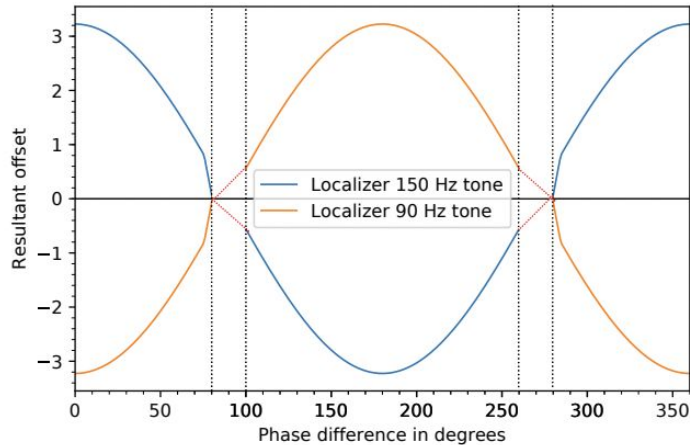


Figure 14: Comparison of calculated offset and the phase difference for localizer

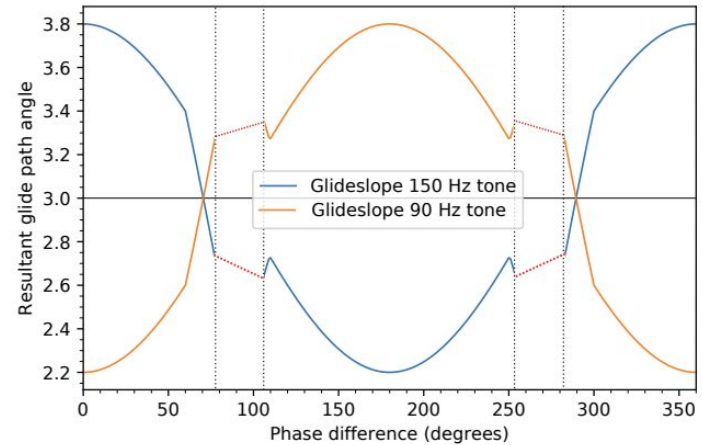


Figure 15: Comparison of calculated offset and the phase difference for glideslope

# Results of Real-Time Amplitude Scaling

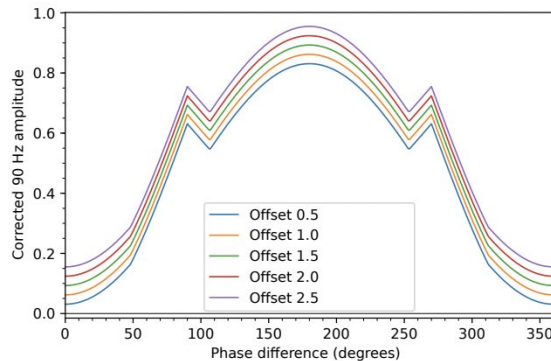


Figure 16: Amplitude scaling algorithm evaluation localizer. Amplitude required to compensate for the effect of phase

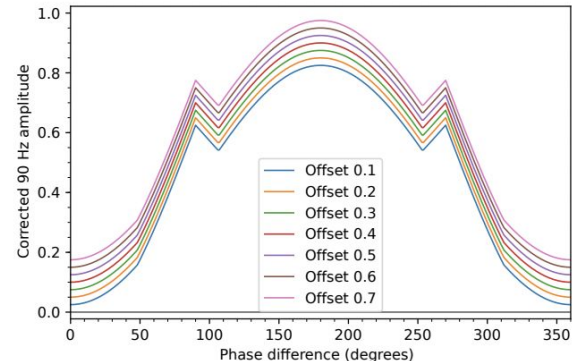


Figure 17: Amplitude scaling algorithm evaluation glideslope. Amplitude required to compensate for the effect of phase

# Power Requirements

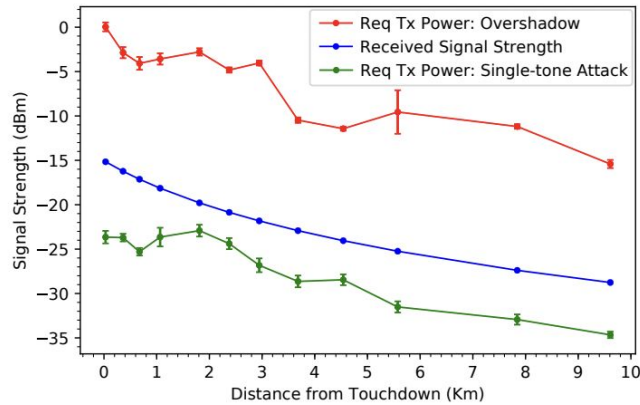


Figure 18: Comparison of required received signal strength for attack methodologies for the localizer

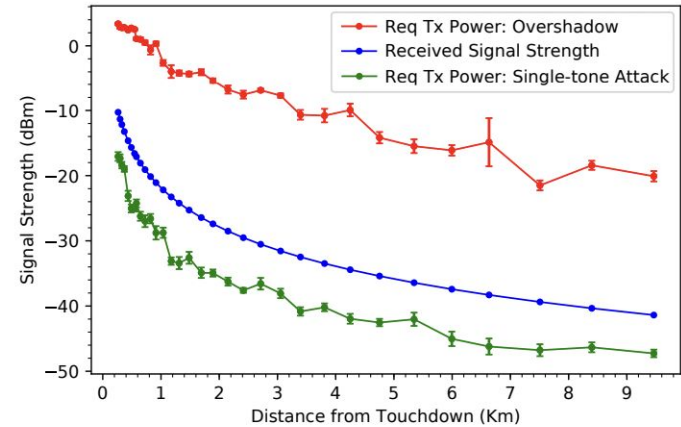


Figure 19: Comparison of required received signal strength for attack methodologies for glideslope

# Acceptance/Rejection

Accept with limitations

# Questions

# Discussion

## Autonomous Commercial Flights

Autonomous systems are already evident in commercial flights with landing and cruise control systems. Would you feel comfortable taking a flight without pilots in the cockpit, or should there always be human intervention as a failsafe?

## ILS

How can ILS be secured or improved? It is demonstrated that the FAA prefers pilots to use instrument based landing rather than visual.



